FAILURE MODES EFFECTS ANALYSIS (FMEA) -- HARDWARE

NUMBER: M0-AG1-E01 -X

SUBSYSTEM NAME: REMOTELY OPERATED FLUID UMBILICAL (ROFU)

REVISION: 12/08/02

PART DATA

PART NAME PART NUMBER
VENDOR NAME VENDOR NUMBER

:ROFU V847-544100-001

:QUICK DISCONNECT MC276-0053-1001/-2001 SYMETRICS (PARKER STRATOFLEX) 516001-1001/-2001

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS: MC276-0053-1001 IS SOCKET, FEMALE HALF, ON ORBITER (TYPE I) MC276-0053-2001 IS NIPPLE, MALE HALF, ON PAYLOAD (TYPE II)

REFERENCE DESIGNATORS: 40P848MD1, 40P848MD2, 40P847MD1, 40P847MD2

QUANTITY OF LIKE ITEMS: 2

TWO MALE HALF QD'S. TWO FEMALE HALF QD'S

FUNCTION:

THE DISCONNECT (QD) PROVIDES THE CONNECTION BETWEEN THE ORBITER PAYLOAD BAY MOUNTED WATER COOLING SYSTEM AND THE PAYLOAD. THE QD CONSISTS OF A PAYLOAD HALF (MALE CONNECTOR) AND AN ORBITER PAYLOAD COOLING SYSTEM HALF (FEMALE). THERE ARE TWO QD'S, ONE EACH FOR COOLANT WATER SUPPLY AND RETURN. THE TWO HALVES ARE ENGAGED BY BEING PUSHED TOGETHER AND DISENGAGED BY BEING PULLED STRAIGHT BACK. THE ENGAGED QD PERMITS FLOW IN EITHER DIRECTION. THE DISENGAGED QD PROVIDES SHUTOFF CAPABILITY. COOLANT SYSTEM PRESSURE IS 300 PSIA. MAXIMUM. THE QD IS REQUIRED TO OPERATE AT ANY SYSTEM PRESSURE FROM 0 TO 300 PSIA. MATED DISCONNECTS PERMIT FLOW RATE OF 1000 LB PER HOUR OF WATER AT 300 PSIA WITH PRESSURE DROP NOT EXCEEDING 1.0 PSI. QD TRAVEL FOR FULL ENGAGEMENT IS 0.984 INCH, MAX. BOTH QD'S ARE ENGAGED OR DISENGAGED SIMULTANEOUSLY BY THE SAME MECHANISM. LINE SIZE IS 5/8 INCH DIAMETER. FLOW RATE IS 500 LB/HR, NOMINAL. THE SYSTEM IS DESIGNED FOR USE OF WATER ONLY AS ITS COOLANT. THE SYSTEM WAS TESTED WITH A PARTIAL CONTAMINATION OF FREON 21 IN CASE THERE MAY BE SOME FREON CONTAIMINATION FROM THE HEAT EXHANGER. THESE TESTS VERIFIED THAT SYSTEM INTEGRITY WOULD BE MAINTAINED WITH THIS LEVEL OF CONTAIMINATION

PAGE 2

PRINT DATE: 2/25/03 DATE/SUPERCEDING: NONE

FAILURE MODES EFFECTS ANALYSIS FMEA -- FAILURE MODE

NUMBER: M0-AG1-E01- 03

REVISION#: 01/23/03

SUBSYSTEM NAME: REMOTELY OPERATED FLUID UMBILICAL (ROFU)

LRU: QUICK DISCONNECT CRITICALITY OF THIS ITEM NAME: QUICK DISCONNECT FAILURE MODE: 2/2

FAILURE MODE:

EXTERNAL LEAKAGE BEFORE OR AFTER DISCONNECT

MISSION PHASE: PL PRE-LAUNCH

LO LIFT-OFF OO ON-ORBIT DO DE-ORBIT

LS LANDING/SAFING

VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA

103 DISCOVERY104 ATLANTIS105 ENDEAVOUR

CAUSE:

VIBRATION, DAMAGED SEAL, DAMAGED OR BROKEN PART IN QD, FREON 21 CONTAMINATION.

CRITICALITY 1/1 DURING INTACT ABORT ONLY: NO

REDUNDANCY SCREEN A) N/A

B) N/A

C) N/A

PASS/FAIL RATIONALE:

A)

N/A

B) N/A

C)

N/A

- FAILURE EFFECTS -

(A) SUBSYSYSTEM:

PAGE 3 PRINT DATE: 2/25/03
DATE/SUPERCEDING: NONE

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NUMBER: M0-AG1-E01-03

THE ACTIVE COOLING SYSTEM CONTAINS LESS THAN 3 GALLONS OF WATER WHEN IT IS NOT CONNECTED WITH MPLM OR 9 GALLONS IF CONNECTED. DEPENDING ON SIZE, ANALYSIS INDICATES THAT, AT THE MOST, APPROXIMATELY 50 IN3 (~ 4 CUPS) OF WATER WOULD BE RELEASED. POSSIBLE LOSS OF ACTIVE COOLING FOR PAYLOAD IN ORBITER PAYLOAD BAY.

(B) INTERFACING SUBSYSTEM(S):

FAILURE IS INDEPENDENT OF ORBITER ECLSS SYSTEM. NO EFFECT ON EQUIPMENT IN PAYLOAD BAY. LEAKED FLUID WOULD FREEZE OR SUBLIMATE ON ORBIT.

(C) MISSION:

POSSIBLE LOSS OF MISSION OBJECTIVE (ISS SUPPORT).

(D) CREW, VEHICLE, AND ELEMENT(S):

NO EFFECT ON CREW OR ORBITER. ANALYSIS INDICATES THAT THE SMALL QUANTITY OF WATER [50 OR SO CUBIC INCHES (~ 4 CUPS)] THAT WOULD BE RELEASED DOES NOT POSE A HAZARD TO THE ORBITER. (REF. ANALYSIS NO. GDS-FSSO-02-012, DATED DECEMBER 12, 02).

(E) FUNCTIONAL CRITICALITY EFFECTS:

N/A

SUCCESS PATHS REMAINING AFTER FIRST FAILURE: 0

- TIME TO EFFECT -

REACTION TIME: IMMEDIATE

-DISPOSITION RATIONALE-

(A) DESIGN:

SEALS ARE STAINLESS STEEL AND NEOPRENE. SEALS ARE RESISTANT TO FREON 21 CONTAMINATION.

(B) TEST:

QUALIFICATION TESTS, REF. CR NO. 60-44-544100-001, INCLUDE RANDOM VIBRATION, SHOCK, LEAKAGE, FLOW AND PRESSURE DROP, SALT FOG, SAND AND DUST, AIR INCLUSION, OPERATIONAL CYCLES AND BURST PRESSURE. A QUICK DISCONNECT EVALUATION TEST WITH FREON 21 SOLUTIONS (AT 1460 PPM FREON 21) VERIFIED THE INTEGRITY OF THE O-RING SEAL.

ACCEPTANCE TESTS INCLUDE PROOF PRESSURE, OPERATIONAL CYCLES, STROKE, OVERSTROKE, ALIGNMENT, FLUID FLOW/PRESSURE DROP. SYSTEM CHECKOUT ON THE VEHICLE CAN BE PERFORMED, BUT LIMITED WHEN MPLM OR OTHER PAYLOAD WITH SIMILAR SIZE INSTALLED, USING GSE MODEL NOS. C73-0012 (ORBITER DISCONNECT SIMULATOR) AND S73-0003 (PAYLOAD DISCONNECT SIMULATOR). GROUND TURNAROUND TEST

PAGE 4 PRINT DATE: 2/25/03
DATE/SUPERCEDING: NONE

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- FAILURE MODE

NUMBER: M0-AG1-E01-03

ANY TURNAROUND CHECKOUT TESTING IS ACCOMPLISHED IN ACCORDANCE WITH OMRSD. SYSTEM CHECKOUT IS LIMITED WHEN MINI PRESSURIZED LOGISTICS MODULE (MPLM) IS INSTALLED

(C) INSPECTION:

RECEIVING INSPECTION

MATERIAL AND PROCESS CERTIFICATIONS ARE VERIFIED BY INSPECTION. ALL PURCHASED PARTS ARE VERIFIED BY INSPECTION.

CONTAMINATION CONTROL

INSPECTION VERIFIES CLEANLINESS IS MAINTAINED AND BURRS REMOVED. VISUAL INSPECTION IS PERFORMED PRIOR TO DELIVERY.

ASSEMBLY/INSTALLATION

DIMENSIONS OF DETAIL PARTS, SURFACE FINISHES, IDENTIFICATION, ASSEMBLY SEQUENCE, INSTALLATION ON SWING ARM AND PAYLOAD ASSEMBLIES ARE VERIFIED BY INSPECTION.

CRITICAL PROCESSES

HEAT TREATMENTS ARE VERIFIED BY INSPECTION.

TESTING

ACCEPTANCE TESTS OF THE QUICK DISCONNECT PER APPLICABLE PROCEDURES ARE VERIFIED BY INSPECTION PRIOR TO DELIVERY. INSPECTION ALSO VERIFIES ACCEPTANCE TESTS OF SWING ARM AND PAYLOAD DISCONNECT ASSEMBLIES WITH QUICK DISCONNECTS INSTALLED.

HANDLING / PACKAGING

HANDLING AND PACKAGING REQUIREMENTS ARE VERIFIED BY INSPECTION

(D) FAILURE

CURRENT DATA ON TEST FAILURES, FLIGHT FAILURES, UNEXPLAINED ANOMALIES, AND OTHER FAILURES EXPERIENCED DURING GROUND PROCESSING ACTIVITY CAN BE FOUND IN THE PRACA DATA BASE.

(E) OPERATIONAL USE:

NONE

- APPROVALS -S&R ENGINEER :D. MAYNE :/S/ D. M.MAYNE :/S/ BOB DUEEASE FOR CARGO/INTEG ITM. :J. CAPALENI DESIGN ENGINEER :L. T. HARPER :/S/ L. T. HARPER SSM :L. J. SALVADOR :/S/ PHAM HOE FOR NASA/DCE :B. BROWN :/S/ B. BROWN :C. STEPHENSON :/S/ C.STEPHENSON_ MOD

PAGE 5 PRINT DATE: 2/25/03 DATE/SUPERCEDING: NONE

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- FAILURE MODE

NUMBER: M0-AG1-E01-03

:/S/ HARRY MALTBY_____ SR&QA USA/SAM

USA CARGO/INTG ELEMENT :S. KUNKEL USA ORBITER ELEMENT S. LITTLE :/S/ S. KUNKEL__

:/S/ SUZANNE LITTLE____